•

- •
- •
- •
- •
- •
- •
- •

Speed Modifications to Improve Efficiency



2002 Annual Convention

Reno, Nevada

Hans D. Neubert ICS 7685 BSME, MSAE, FAA Structures DER PA-30 N7331Y

Copyright 2002 - Hans D. Neubert

Seminar Presentation Topics

- Drag Basics
- Drag Reduction Fundamentals
- Oil Smear Testing Performed
- Tuft Testing Performed
- Speed Modifications Incorporated
- Propeller Efficiency
- Cooling Drag
- Cowlings
- Conclusions/Recommendations

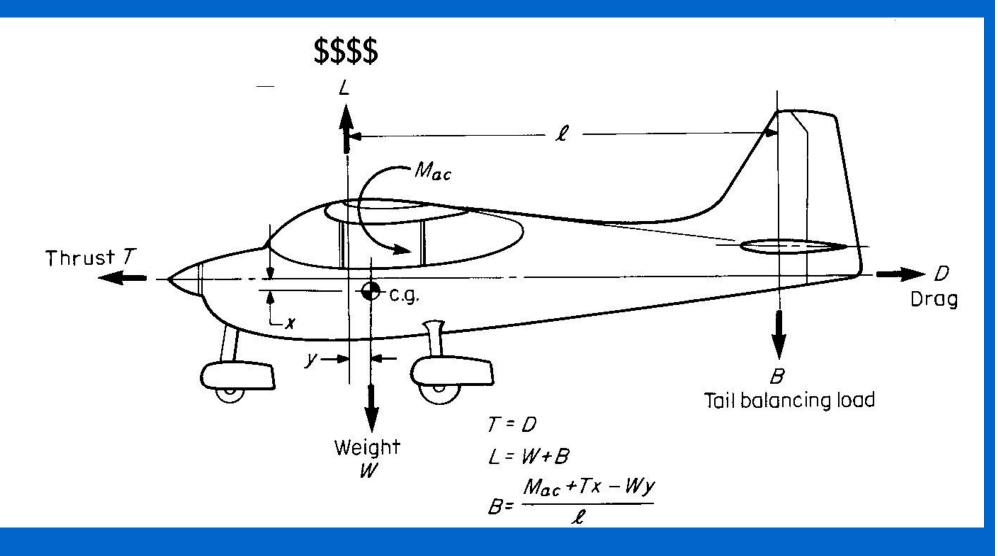
Definitions

- Dynamic Pressure
 - Air Density times Velocity Squared
- Reynolds Number
 - Velocity times Characteristic Length divided by Kinematic Viscosity of air
- Drag Component
 - Drag equals Dynamic Pressure times Drag Coefficient times Characteristic Dimension

Drag Components

- Total Drag equals Parasite Drag plus Induced Drag
- Parasite Drag has 5 Main Components
 - Skin Friction Drag
 - Pressure Drag (or Form Drag)
 - Interference Drag
 - Trim Drag
 - Cooling Drag
- Induced Drag
 - Results from the generation of lift
 - Increases with altitude and during climb

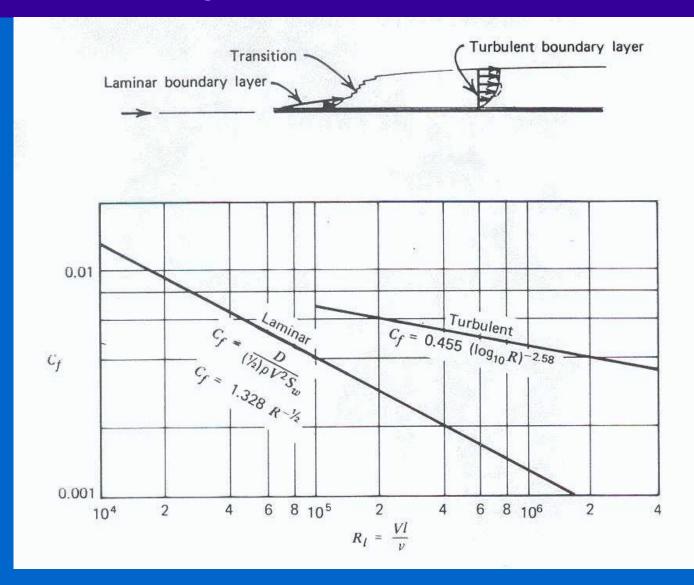
Aircraft in Static Equilibrium - Level Flight



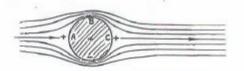
Balance of Thrust and Drag

- In Steady, Level Flight Thrust equals Drag
- Thrust equals Hp times Prop Efficiency divided by Velocity (corrected for units)
- Drag equals all drag components added together. Parasite Drag can also be expressed as an equivalent drag area having a drag coefficient of one multiplied by the wing area
- Units are: pounds thrust = pounds drag

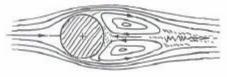
Skin Friction Drag



Pressure (Form) Drag

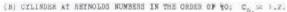


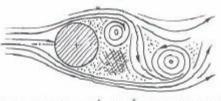
(A) FLOW PATTERN OF CIRCULAR CYLINDER IS NON-VISCOUS FLOW; NO DRAG.



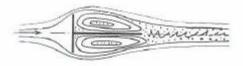
Form drag on shapes is very difficult to predict. Wind tunnel testing is the usual method for obtaining drag coefficient vs. Reynolds number.

Form drag is the integral of pressure over the entire surface area.

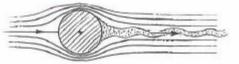




(c) CYLINDER BETWEEN $B_{\rm d}$ = 10^4 and 10^5_1 VORTER STREET with $D_{\rm m}$ = 1.2.



(D) PLATE WITH "SPLITTER" DEVICE IN MAKE; C1. = 1.6.



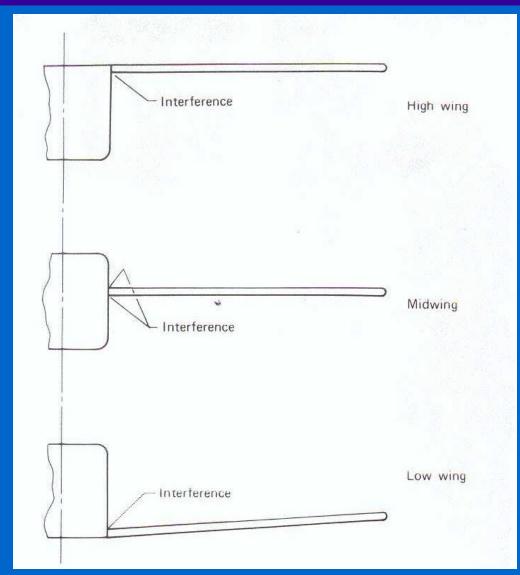
(6) OYLINDER ABOVE CRITICAL REYNOLDS NUMBER WITH $C_{p_1} = 0.3$.



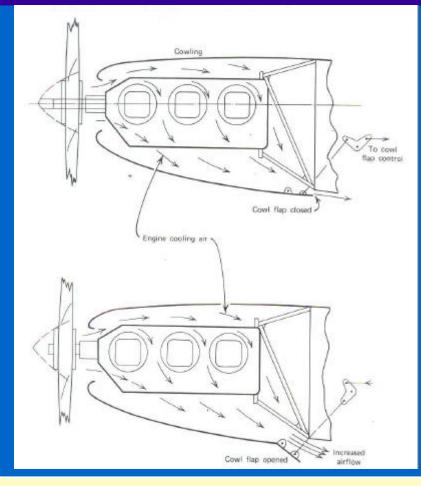
(F) STREAMLINE SECTION WITH CD. IN THE ORDER OF 0.06.

Interference Drag

When two shapes intersect or are in proximity, their pressure distribution and boundary layers can interact, resulting in a net combined drag that is higher than the sum of the separate drags.



Cooling Drag



Jutside air enters upper plenum, air velocity converts to increased pressure, air absorbs eat from cylinder fins, and heated, higher pressure air leaves lower plenum creating

Copyright 2002 - Hans D. Neubert

Estimates of Stock Comanche Total Drag

	Total Drag Breakdown		
ltem	250/260 Single		Twin
Total Drag	420 lb		500 lb
Induced Drag	30 lb		40 lb
Parasite Drag	390 lb		460 lb
Equiv. Flat Plate Area	4.53 sq.ft.		4.46 sq.ft.
Airframe Drag (Skin friction)	245 lb		265 lb
Airframe Drag (Form)	54 lb		60 lb
Interference Drag	42 lb		68 lb
Cooling Drag	39 lb		55 lb
Trim Drag	10 lb		12 lb
Assumptions:			
Max. Speed at SL, Max. HP, Aft CG.			
Gross Weight, Stock Configuration			

۲

What Can You Do to Reduce Parasite Drag

- Skin Friction Drag
 - Essentially nothing Keep Leading Edge clean to maintain Laminar Flow
- Pressure (Form) Drag
 - Essentially nothing Controlled by airplane shape & configuration. Some speed mods deal with Form Drag.
- Interference Drag
 - Judicious application of Speed Mods & airflow enhancements
- Trim Drag
 - Maintain Aft CG using payload or baggage weight
- Cooling Drag
 - Maintain seal between upper and lower plenums no leaks

Prior to Any Speed Modifications

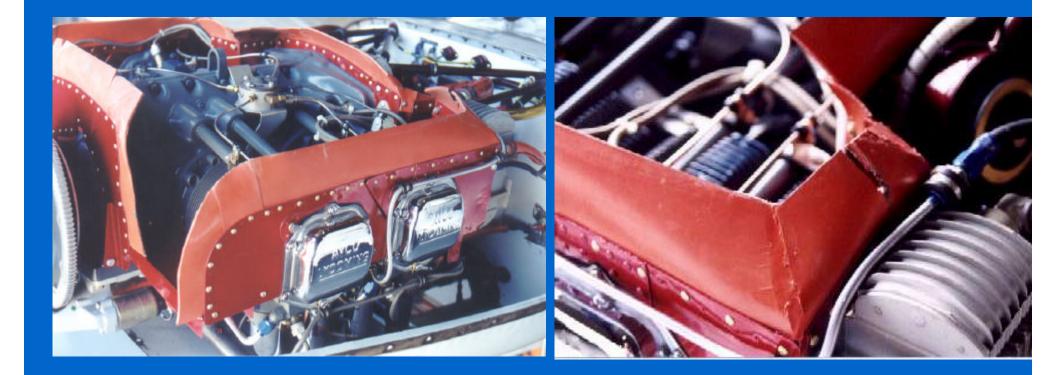
- Rig Airplane for Maximum Speed
- Verify Gear Retraction and doors per Service Manual
- Seal Cowling to Minimize Internal Leaks
- Remove Unnecessary Antennas and Protuberances
- Service Propeller for Nicks and Dings Paint Backside

Gear Door Adjustment Test



9 lb. weight at center of gear door -1/8" door deflection

Engine Baffle Seal

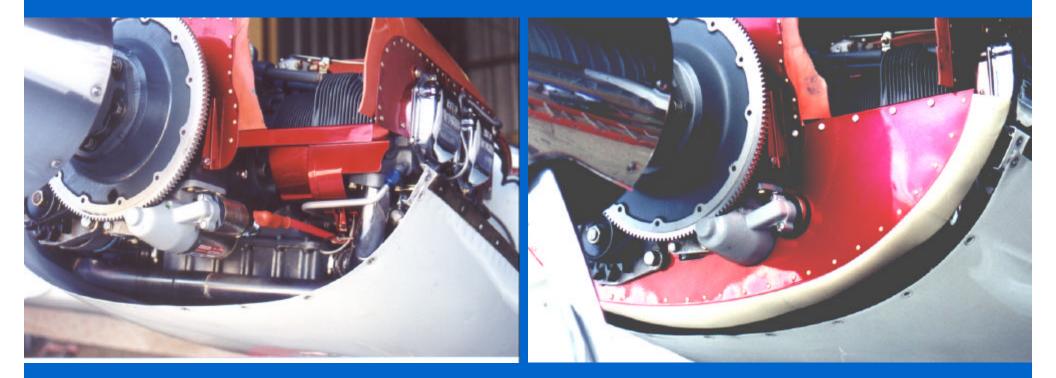


Copyright 2002 - Hans D. Neubert

Engine Baffle Seal



Maintain Cowl Airflow in Aft Direction



Before - Stock Configuration

After - Added Lower Seal

Copyright 2002 - Hans D. Neubert

Things I Have Done to Minimize Drag

- Oil Smear Studies
 - Mix Engine Oil with Fine Powder (Use Two Colors)
 - Dab on Plane in Hex Pattern
 - Take Photos Before and After Short Flight
- Tufting Studies
 - Contrasting Yarn Taped to Plane
 - Take Photos (Video or Still) From Cabin or Chase Plane
- Eliminated Sharp Edged Airflow Areas
 - Engine Induction Air Intake
 - Cowl Exit Air Outlet
- Fab'd or Purchased STC'd Airflow Enhancement Fairings

Twin Heater Exhaust





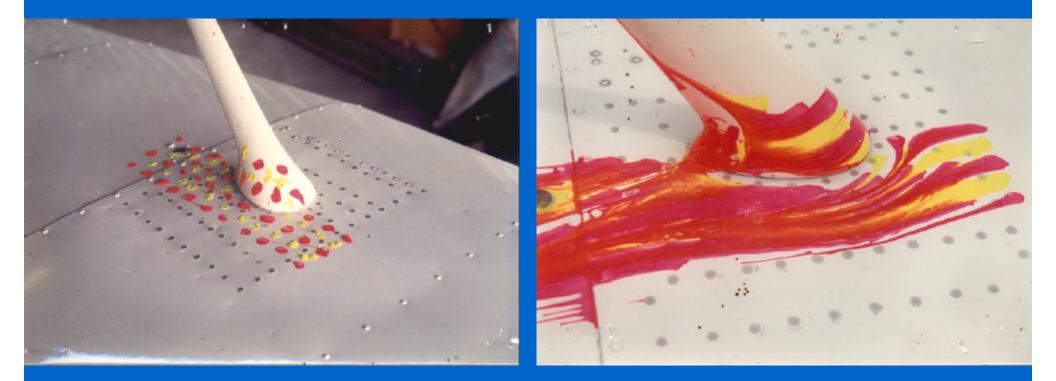
Copyright 2002 - Hans D. Neubert

Heater Exhaust Fairing



Copyright 2002 - Hans D. Neubert

Blade Comm Antennas



Blade Comm Antennas



Copyright 2002 - Hans D. Neubert

Blade Comm Antenna Fairing





Copyright 2002 - Hans D. Neubert

Wing Root and Door

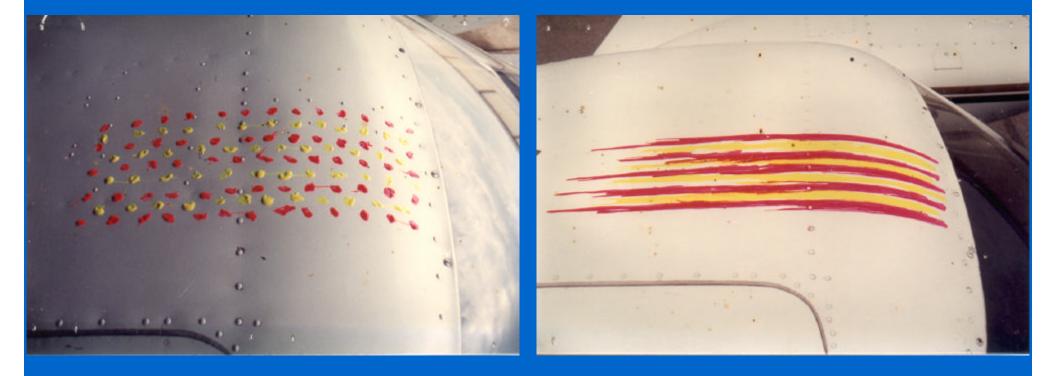


Wing Root Fairing



Copyright 2002 - Hans D. Neubert

Windshield - Cabin Junction



Airflow over windshield to cabin roof appears normal - Vortex generators not req'd

Copyright 2002 - Hans D. Neubert

Wing - Fuselage Trailing Edge



Adverse airflow not observed - No action taken

Copyright 2002 - Hans D. Neubert

Airflow on Nacelle



Copyright 2002 - Hans D. Neubert

Wing at Stall - 18" & 2100 RPM



Copyright 2002 - Hans D. Neubert

Wing in Landing Configuration (15° Flaps)

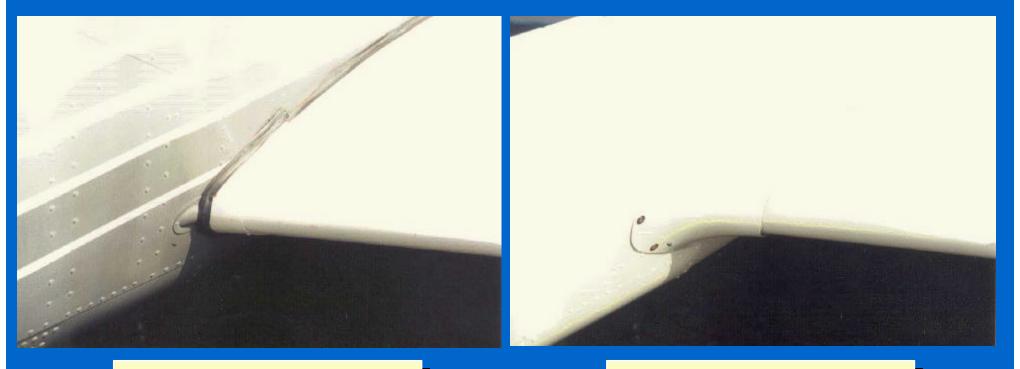


Copyright 2002 - Hans D. Neubert

Purchased STC'd Speed Modifications

- Consider Speed Modifications based on value and historical customer satisfaction
- Value = Buck Spent per MPH gained (my criteria)
 - \$100 \$200 / mph gained = good deal
 - \$200 \$300 / mph gained = acceptable deal
 - \$300 \$400 / mph gained = marginal deal
 - > \$400 / mph gained = unacceptable deal
- Keep in mind speed modification claims for speed gain are not additive - As interference and form drag are reduced and plane goes faster, skin friction and pressure drag increase, resulting in a smaller, if any, net gain.

Wing Root Fairing



Stock Configuration

K2U Wing Root Fairing

Remove Things that are Round





Standard Grimes Strobe

K2U Fin Cap - Strobe in

Rudder

Improved Airflow Wing Tips





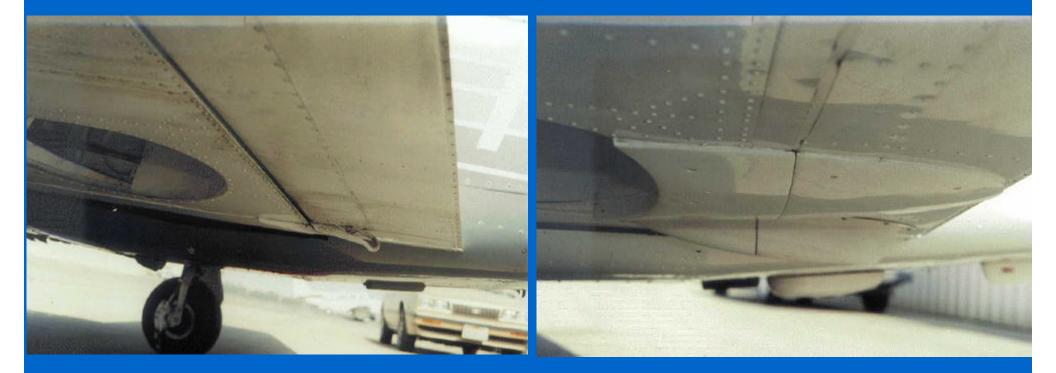
MetCoAir Wingtips with External

ΨĽ¥ght

Johnston Wingtips with Enclosed

The second seco

Landing Gear Afterbody Fairings



Stock Configuration

K2U Gear Lobe shown Lopresti Speed Spats similar

Copyright 2002 - Hans D. Neubert

Flap and Aileron Gap Seals



Stock Configuration

Copyright 2002 - Hans D. Neubert



K2U Flap & Aileron Seals

Flap Track Bracket Fairings



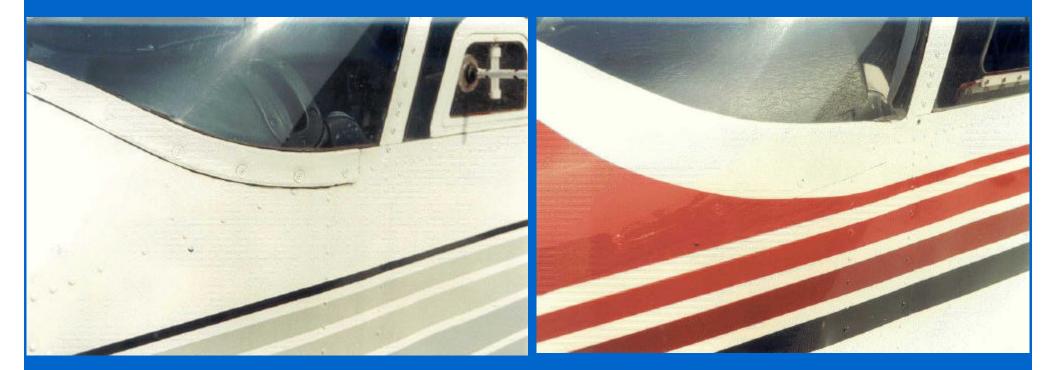


Stock Configuration

Lopresti Speed Splitters

Copyright 2002 - Hans D. Neubert

Windshield Cowler



Stock Configuration

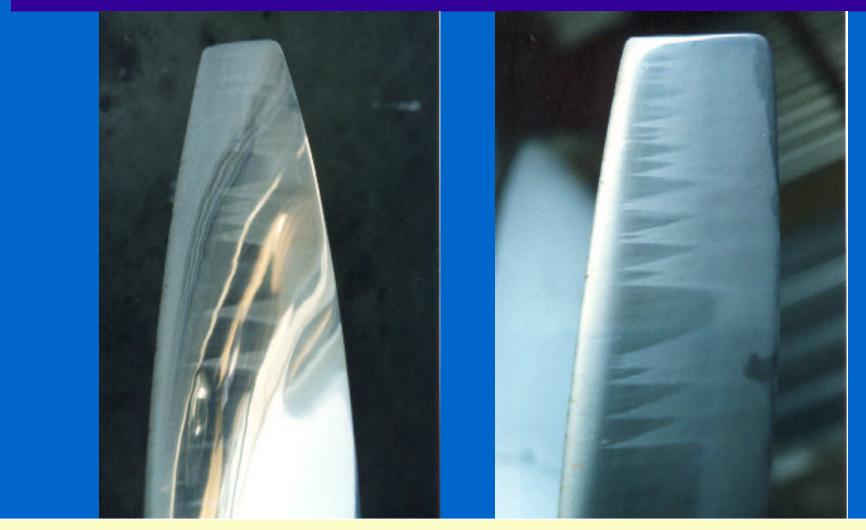
K2U Windshield Cowler

Brake Reversal



Requires a cutout on rib at WS 33.06. Cutout reinforced with Doubler + Stiffener. Swap wheel and brake caliper Left to Right, and re-route brake line. Split hubcap is optional.

Propeller Cleanliness for Maximum Performance



Pictures show effect of laminar vs. turbulent flow caused by leading edge

Copyright 2002 - Hans D. Neubert

Nose Bowl - Singles







Nose Bowl - Twin



Stock Twin Inlet = 64 sq. in.

Lopresti Twin Inlet = 38 sq. in.

Similarity of Twin to Nemisis Cowl



Long Prop Extension - Angled Air Inlets - Controlled Expansion into Cowl Plenum -Complete Seal around Engine

Copyright 2002 - Hans D. Neubert

Nosebowl Idea That Did Not Work

Reduce Inlet Area by 20 sq. in.



Faired Inlet - Foam Expansion Ramp Shape Internal

Copyright 2002 - Hans D. Neubert

Cowl Internal Airflow - Induction Velocity Stack



Induction Inlet - Stock

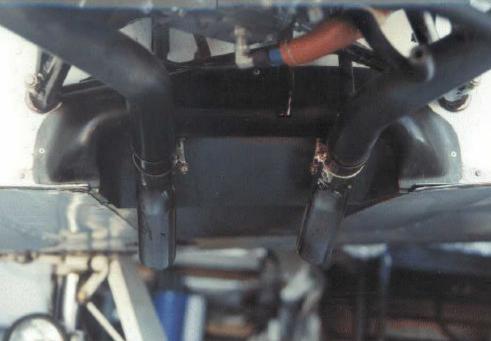
Induction Inlet with Velocity

Stack

Copyright 2002 - Hans D. Neubert

Cowl Internal Airflow - Exit Air Fairing





Exit Tunnel - Stock

Firewall Exit Fairing with K2U Exhaust Fairing

Copyright 2002 - Hans D. Neubert

New Twin Nosebowl - Neubert Version



New Twin Nosebowl - First Flight



Copyright 2002 - Hans D. Neubert

Measured Cowl and Engine Temperatures

			Temperature °F	
Location		Cowl Door Open		Cowl Door Closed
Outside Air Temperature	61°F			
Air Temperature at Induction Inlet		70 °F		73 °F
Cylinder Head Temperature (Average)		360 °F		403 °F
Left Magneto (at Data Plate)		147 °F		152 °F
Right Magneto (at Data Plate)		129 °F		139 °F
Vacuum Pump (at Data Plate)		168 °F		174 °F
Fuel Pump (on Stem)		161 °F		174 °F
Exit Air at Cowl Door		140 °F		158 °F
Conditions:				
5000 Ft, 24 Square, 8.2 GPH, 165 KIAS				

New Nosebowl - Flight #1

•

Speed Mods in Action



Antenna Fairing, Gear Lobes, Flap Track Fairing, Ailerons Reflexed Up, Exhaust Pipe Ext.

Copyright 2002 - Hans D. Neubert

Recommendations Incorporate Low Cost Speed Mods First

- Perform initial rigging, gear check, engine seal first
- Brake Reversal with Hub Caps on Dual Fork Models
- Reflex Ailerons up 3 Degrees (see Tips Special)
- Blade Nav and Comm Antennas in lieu of wire
- Digital Thermometer with small probe
- Aileron + Flap Gap Seals
- Eliminate round things (if possible)
- Maintain weight in baggage compartment for aft CG

Summary

- Keep in mind speed modification claims for speed gain are not additive - As interference drag is reduced and plane goes faster, skin friction and pressure drag increase, resulting in a smaller, if any, net gain.
- Keep in mind that aircraft performance and engine power are greatly influenced by density altitude.
- Select Speed Mods based on value.
- Keep propeller and wing leading edge clean at all times.
- Pay attention to detail.
- Be realistic in your expectations.